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Some asymptotic results for bifurcations in pure bending of rubber blocks

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Abstract

The bifurcation of an incompressible neo-Hookean thick hyperelastic plate with a ratio of thickness to length η and subject to pure bending is considered within a plane-strain framework. The two incremental equilibrium equations corresponding to a nonlinear pre-buckling state of strain are reduced to a fourth-order linear eigenproblem that displays a multiple turning point. It is found that for $0 < \eta < \infty$ the plate experience an Euler-type buckling instability which in the limit $\eta \rightarrow \infty$ degenerates into a surface instability. Singular perturbation methods enable us to capture this transition, while direct numerical simulations corroborate the analytical results.

Keywords: hyperelastic plates, incremental equations, turning points, boundary layers.

1 Introduction

The development of compressive stresses in mechanical structures is well known to be responsible for Euler-type buckling instabilities. What is less recognised is that such scenarios are likely to occur in a number of cases that, apparently, are of a completely different nature. A typical example is the phenomenon of stress concentration in perforated thin elastic plates subjected to tension. Usually, the holes act as stress concentrators that can be completely or only partially surrounded by compressed regions. If the pulling forces are sufficiently strong an out-of-plane bending instability is experienced locally near the sites of the holes. A systematic investigation of problems of this nature has recently been initiated by Coman *et al.* [1, 2, 3, 4].

A second example where Euler-type buckling is indirectly encountered is provided by the pure bending of a thin and short elastic tube. The curved configuration adopted by the tube is characterised by compressive axial stresses on the concave side, whereas tension will prevail on the convex part. Experience shows that a regular instability pattern consisting of many little ripples will develop along the former region, eventually leading to the creation of one or several kinks that signal the collapse of the tube.

The stability problem of pure bending in thin-walled tubular structures has a long history and there is a vast mechanical engineering literature dealing with various aspects; some of it is aptly summarised in the authoritative account of Kyriakides & Corona [5]. On the mathematical side, noteworthy contributions in the present context are the works by Seide & Weingarten [6] and those by Tovstik *et al.* (briefly summarised in [7]). The former investigation is based on the Donnell-von Kármán buckling equations linearised around a variable-coefficient membrane state of stress. The resulting boundary value problem was analysed numerically with the help of the Galerkin method, and it was found that the circumferential shape of the buckled cylinder displays a small dimple on the compressed side. Several versions of the same problem have been re-considered in [7] from the point of view of asymptotic analysis. Both works just now mentioned made the simplifying assumption that the rippling pattern is the same at every point along the axis of the cylinder or, in other words, a solution with separable variables was *a priori* postulated. The assumption is sensible for short tubes (which are fairly stiff), but it is inadequate for modelling the collapse in the elasto-plastic regime for moderate lengths, which turns out to require a very different approach (cf. [5]).

Our main aim in the present investigation is to re-visit the pure bending of a rubber block deforming in plane strain, a problem that has several points in common with the tube bending mentioned above. Unfortunately, the literature in this area has focused mainly on describing the deformation itself rather than its potential bifurcations. The typical scenario is outlined in Figure 1: the undeformed configuration is shown in the left-hand sketch and is characterised by the geometric parameters $2L$ (*length*), H (*height*), and $2A$ (*thickness*); the deformed block appears on the right in the same Figure. The plane-strain hypothesis simplifies the problem considerably, since one needs deal only with cross-sections (shown shaded) perpendicular to the vertical axis of the block and situated sufficiently far away from the lower and upper faces.

Pioneering work on pure bending stability was carried out by Triantafyllidis [8], who examined incremental bifurcation equations for a couple of piecewise power law constitutive models, including a hypoelastic one. He pointed out that the underlying instability mechanism is a surface instability similar to that encountered in the plane-strain half-space problems discussed by Hill & Hutchinson [9] or Young [10]. Haughton [11] performed a similar analysis for hyperelastic materials in a three-dimensional context (neo-Hookean, mostly) and allowed for vertical compression as well. The instability was found to be of Euler-type but his interpretation of some of the results is wrong (as we shall explain in §3). A novel feature in [11] is the interac-

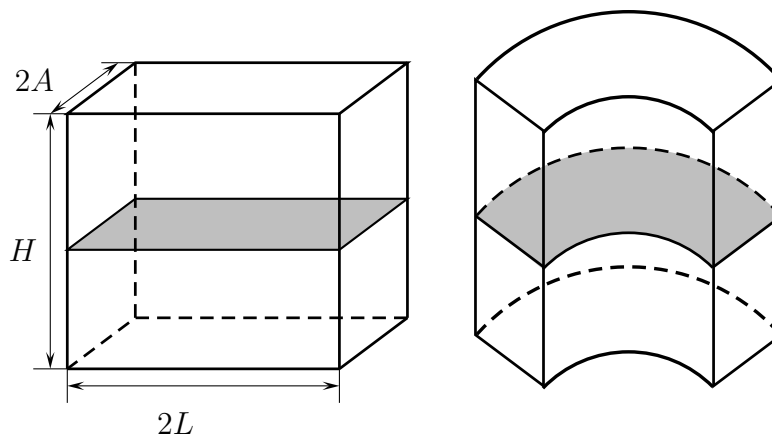


Figure 1: Cylindrical bending of a rubber block; left: *reference* configuration, right: *current* configuration. Shaded areas indicate two generic cross-sections perpendicular to the vertical axis.

tion between two different modes of instability, one due to pure bending, the other related to compression. Dryburgh & Ogden [12] introduced thin coatings on the curved boundaries of the bent block and made comparisons with the uncoated case. Their findings show that, relative to the latter case, bifurcation is generally promoted by the presence of surface coating, on either or both curved boundaries, that is the bifurcation occurs at smaller strains. The relative sizes of the shear moduli for the coating and, respectively the bulk material was found to play an important role in describing this phenomenon.

The finite elasticity works reviewed above share a common feature in that they all deal with incompressible materials. So far little is known about the role played by compressibility on the bifurcation behaviour in pure bending. The reason might be rooted in the absence from the literature of a manageable closed-form expression for the pre-bifurcation deformation. Aron & Wang [13, 14] touched upon issues like existence and uniqueness for bending deformations in unconstrained elastic materials, while Timme *et al.* [15] used Hencky's compressible elasticity model to investigate closed-form solutions for cylindrical bending. They succeeded in deriving explicit expressions for the bending angle and moment in terms of the circumferential stretches on the curved boundaries. The solution is quite involved and it seems unlikely to be useful for anything but numerical calculations. Moreover, the particular Hencky elasticity framework is restricted by moderate deformations only.

A critique of bifurcation phenomena in pure bending was given by Gent & Cho [16] who pointed out that their experiments did not agree with the theoretical predictions based on the surface-instability concept proposed in [8]. In particular, they found that the instability occurs for a smaller degree of compression and the block adopts a configuration with a small number of *sharp creases* on the inner surface. To fully explain this observation would require a nonlinear *post-buckling* analysis because the bifurcation involved is probably of *subcritical* type. We note in passing that Gent & Cho's creases are, to a certain extent, very similar to those encountered on the curved surface of severely torsioned stocky rubber cylinders [17]. These phenomena are likely to be related to the failure at the boundary of the *complementing condition* (see [18] and the reference therein), and they fall outside the scope of our study.

With this background in mind, we shall re-consider in the next sections a particular instance of the pure bending problems taken up in [11, 12]. The aim is to elucidate the nature of the

instability and to analyse the mathematical structure of the governing boundary value problem when $\eta \equiv A/L \gg 1$. To avoid “missing the forest for the trees”, the model investigated will be confined to incompressible neo-Hookean materials. In §2 these assumptions are showed to yield an eigenproblem for a fourth-order partial differential equation with variable coefficients, subsequently simplified by seeking a solution with separable variables. Direct numerical simulations reveal an Euler-type buckling phenomenon for $0 < \eta < \infty$, but in the limit $\eta \rightarrow \infty$ this degenerates into a surface instability. Some erroneous interpretations proposed by previous investigators (e.g., [8] or [11]) are also corrected here for the first time. As demonstrated in §4, the transition regime between the two different forms of instabilities can be efficiently captured by singular perturbation methods. The two contrasting asymptotic methods employed are discussed separately in §4.1 (WKB) and, respectively, §4.2 (boundary layers). The former would seem to be the most appropriate because the differential equation in question has variable coefficients. However, it transpires that a conventional boundary-layer analysis sheds more light and helps us to steer clear from the confusion created by the presence of a *multiple turning point*. Unlike in the recent studies [1, 2, 3, 4], here turning points play no role whatsoever (the same is true for the related works [25, 27]). The paper concludes with a discussion of the results obtained, together with suggestions for further study.

2 Overview of the model

Finite pure bending of incompressible hyperelastic materials is discussed in a number of books like, for example, [19] or [20]. To make the paper reasonably self-contained, we summarise some of those ideas below.

The reference configuration of the initially undeformed rectangular cross-section of the hyperelastic block is the region (see Figure 2)

$$\mathcal{B}_R := \{(X_1, X_2) \in \mathbb{R}^2 \mid -A \leq X_1 \leq A, -L \leq X_2 \leq L\} .$$

Supposing that the block is bent (symmetrically with respect to the x_1 -axis) into a sector of circular cylindrical tube, the current configuration of the deformed cross-section is easily represented in polar co-ordinates by the domain

$$\mathcal{B}_C := \{(r, \theta) \in \mathbb{R} \times (0, 2\pi] \mid -r_1 \leq r \leq r_2, -\omega_0 \leq \theta \leq \omega_0\} .$$

Rivlin [21] showed that for an incompressible elastic material this type of deformation may be described by the mapping

$$r = (d + 2X_1/\omega)^{1/2}, \quad \theta = \omega X_2, \quad (1)$$

where d is a quantity determined by the particular constitutive law adopted and ω can serve as a control parameter as it is related to the *angle of bending*, $\omega_0 := \omega L$. Since the plate cannot be bent into itself, we should require that

$$0 < \omega_0 \leq \pi, \quad (2)$$

an assumption used tacitly henceforth. Although the deformation recorded in (1) seems to have an iconic status among workers in elasticity, it is clear that the kinematics afforded by that expression are somewhat restricted. The lines $X_1 = \text{const.}$ become arcs of the circle $r = \text{const.}$, while the lines $X_2 = \text{const.}$ are transformed in lines $\theta = \text{const.}$; in other words, “cross-sections”

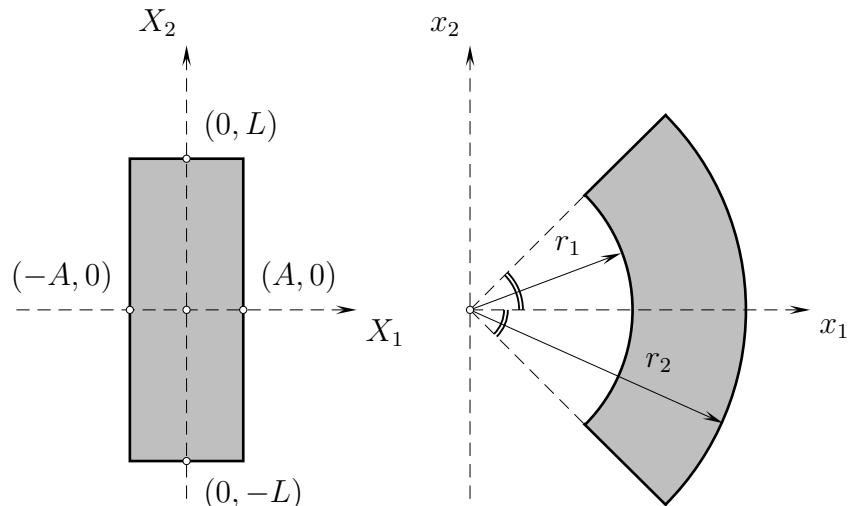


Figure 2: The *undeformed* (left) and *deformed* (right) cross-sections of the rubber block shown in Figure 1. Bending is symmetric with respect to the x_1 -axis so that the two angles marked are congruent and equal to ω_0 ; see the text for more details.

perpendicular to the vertical symmetry axis of \mathcal{B}_R , remain orthogonal to the deformed axis of the current configuration \mathcal{B}_C . This is somewhat at odds with the commonly accepted point of view in structural mechanics, according to which pure bending of thick sandwich panels (e.g., [24]) is based on models that allow cross-sections to slide relative to the normal to the deformed axis. Nonetheless, the *nonlinear* mapping (1) is still a sensible choice for the type of questions we want to answer, at least in a first approximation.

The bifurcation analysis carried out in this work is based upon linearising the plane-strain equations of finite elasticity around the *nonlinear* pre-buckling deformation (1). This approach to linearised or incremental bifurcations is well established (see [22, 20], for example), so we shall not rehearse it here. Instead, we limit ourselves to pointing out the key steps that lead to our eigenproblem.

We shall assume that the constitutive behaviour of the material is characterised by a strain-energy function $W \equiv W(\lambda_r, \lambda_\theta)$, where the principal stretches λ_r and λ_θ are associated with the Eulerian principal directions \mathbf{e}_r and, respectively, \mathbf{e}_θ . Due to the incompressibility constraint these can be written as

$$\lambda_r = \lambda^{-1} \quad \text{and} \quad \lambda_\theta \equiv \lambda := \omega r,$$

which defines the notation λ . According to [20], the two-dimensional version of the incremental equations of equilibrium for incompressible elasticity read

$$\operatorname{div} \mathring{\mathbf{s}} = 0, \quad \operatorname{div} \mathring{\mathbf{u}} = 0, \quad (3)$$

where $\mathring{\mathbf{u}} = (\mathring{u}(r, \theta), \mathring{v}(r, \theta))$ is the incremental displacement field, and $\mathring{\mathbf{s}}$ denotes the incremental nominal stress tensor with components

$$\mathring{s}_{ij} = L_{ijkl} \mathring{F}_{kl} + p \mathring{F}_{ij} - \mathring{p} \delta_{ij}, \quad i, j \in \{r, \theta\}.$$

Here \mathring{p} is the increment in the Lagrange multiplier $p \equiv p(r, \theta)$ (the “hydrostatic pressure”),

while \mathring{F}_{ij} represent the components of the incremental deformation gradient,

$$\mathring{\mathbf{F}} = \begin{bmatrix} \mathring{u}_{,r} & \frac{1}{r}(\mathring{u}_{,\theta} - \mathring{v}) \\ \mathring{v}_{,r} & \frac{1}{r}(\mathring{u} + \mathring{v}_{,\theta}) \end{bmatrix}.$$

Finally, L_{ijkl} are the components of the fourth-order tensor of instantaneous incremental moduli which, in Eulerian principal axes has 15 independent non-zero such components (cf. [20])

$$L_{iijj} = \lambda_i \lambda_j W_{ij}, \quad (4a)$$

$$L_{ijij} = \begin{cases} \frac{\lambda_i^2(\lambda_i W_i - \lambda_j W_j)}{\lambda_i^2 - \lambda_j^2} & \text{if } i \neq j, \lambda_i \neq \lambda_j, \\ \frac{1}{2}(L_{iiii} - L_{iijj} + \lambda_i W_i) & \text{if } i \neq j, \lambda_i = \lambda_j, \end{cases} \quad (4b)$$

$$L_{ijji} = L_{jijj} = L_{ijij} - \lambda_i W_i, \quad (4c)$$

with $W_i \equiv \partial W / \partial \lambda_i$, $W_{ij} \equiv \partial^2 W / \partial \lambda_i \partial \lambda_j$, and the summation convention does not apply.

Direct calculations show that the system of equations (3) can be reduced to

$$\begin{aligned} r^2 \mathring{p}_{,r} = r [r(L'_{1111} - L'_{1122} + p_{,r}) + L_{1111} + L_{2222} - 2L_{1122}] \mathring{u}_{,r} \\ + r^2(L_{1111} - L_{1122}) \mathring{u}_{,rr} + L_{2121}(\mathring{u}_{,\theta\theta} - \mathring{v}_{,\theta}) + rL_{2112} \mathring{v}_{,r\theta}, \end{aligned} \quad (5)$$

$$\begin{aligned} r \mathring{p}_{,\theta} = (rL'_{1212} + L_{1212})(r \mathring{v}_{,r} + \mathring{u}_{,\theta} - \mathring{v}) + r^2 L_{1212} \mathring{v}_{,rr} \\ + r(L_{2112} + L_{1122} - L_{2222}) \mathring{u}_{,r\theta}. \end{aligned} \quad (6)$$

To avoid overdoing the notation we have used the correspondence $r \rightarrow 1$ and $\theta \rightarrow 2$ for the incremental moduli, and have indicated their derivatives with respect to r by dashes. A further simplification is afforded by the incompressibility condition which allows us to deduce the existence of a potential $\phi \equiv \phi(r, \theta)$ such that

$$\mathring{u} = \frac{1}{r} \frac{\partial \phi}{\partial \theta}, \quad \mathring{v} = -\frac{\partial \phi}{\partial r}. \quad (7)$$

The upshot of this observation is that the two equations (5,6) can now be combined into a single partial differential equation for the potential function. After some routine (but lengthy) manipulations, we end up with

$$\sum_{j=1}^4 \mathcal{L}_j[\phi] = 0, \quad (8)$$

with \mathcal{L}_j partial differential operator of the j -th order defined according to

$$\begin{aligned} \mathcal{L}_4 &:= \alpha r^4 \frac{\partial^4}{\partial r^4} + 2\beta r^2 \frac{\partial^4}{\partial r^2 \partial \theta^2} + \gamma \frac{\partial^4}{\partial \theta^4}, \\ \mathcal{L}_3 &:= 2r^3 (r\alpha)' \frac{\partial^3}{\partial r^3} + 2r^3 \left(\frac{\beta}{r}\right)' \frac{\partial^3}{\partial r \partial \theta^2}, \\ \mathcal{L}_2 &:= r^4 \left[\alpha'' + \left(\frac{\alpha}{r}\right)' \right] \frac{\partial^2}{\partial r^2} - r^2 \left[\alpha'' + \left(\frac{\alpha + 2\beta}{r}\right)' - \frac{\gamma}{r^2} \right] \frac{\partial^2}{\partial \theta^2}, \end{aligned}$$

$$\mathcal{L}_1 := -r^3 \left[\alpha'' + \left(\frac{\alpha}{r} \right)' \right] \frac{\partial}{\partial r},$$

and

$$\alpha(r) := L_{1212}, \quad \gamma(r) := L_{2121}, \quad \beta(r) := \frac{1}{2}(L_{1111} + L_{2222}) - (L_{1122} + L_{2112}).$$

The form (8) of the bifurcation equation is valid for any choice of incompressible hyperelastic material but, as it stands, the model is not easily amenable to analytical work. Before further simplifications are implemented, we must address the issue of boundary conditions for (8).

The two curved boundaries of \mathcal{B}_C are taken to be traction-free, a constraint which demands

$$\alpha r^3 \phi_{,rrr} - (2\beta + \alpha)(\phi_{,\theta\theta} - r\phi_{,r\theta\theta}) = 0, \quad \text{for } (r, \theta) \in \{r_1, r_2\} \times (0, 2\pi], \quad (9a)$$

$$\phi_{,\theta\theta} + r\phi_{,r} - r^2\phi_{,rr} = 0, \quad \text{for } (r, \theta) \in \{r_1, r_2\} \times (0, 2\pi]. \quad (9b)$$

These conditions can be obtained as a particular case of the calculations of Dryburgh & Ogden [12], to which the reader is referred for more information.

Next, we look for separable solutions of the bifurcation equation (8) in the form

$$\phi(r, \theta) = \Phi(r) \cos(m\theta), \quad (10)$$

where $m \in \mathbb{N}$ is the azimuthal mode number related to the number of ripples on the compressed side of the rubber block, while $\Phi(r)$ is the infinitesimal amplitude of this cosine rippling pattern. Several types of boundary conditions are possible for the straight boundaries of \mathcal{B}_C . Following [11, 12] we consider zero incremental displacement in the radial direction and vanishing normal traction. It can be shown that these conditions are satisfied as long as

$$m = \frac{n\pi}{\omega L}, \quad (11)$$

for some positive $n \in \mathbb{Z}$. With this information in hand, all that remains to be done is to choose a constitutive model and carry out the simplification of (8) with the help of the assumed form solution recorded in (10).

The bulk material is modelled by a simple neo-Hookean strain-energy function specialised to plane-strain elasticity,

$$W(\lambda_r, \lambda_\theta) = \frac{1}{2}\tau(\lambda_r^2 + \lambda_\theta^2 - 2),$$

τ being the ground-state shear modulus of the material. As shown by Rivlin [21] and subsequently discussed by others [8, 11, 12], for this particular choice of constitutive law the constant d in (1) is determined by

$$d = \frac{L^2}{\omega_0^2}(1 + 4\eta^2\omega_0^2)^{1/2}.$$

On making use of (10) in (8) results in an ordinary differential equation which, when expressed in the non-dimensional variables

$$\rho := \frac{r}{L}, \quad \mu := n\pi, \quad \eta := \frac{A}{L},$$

can be cast as

$$\Phi'''' + \mathcal{P}(\rho)\Phi''' + \mathcal{Q}(\rho)\Phi'' + \mathcal{R}(\rho)\Phi' + \mathcal{S}(\rho)\Phi = 0, \quad \text{on } \rho_1 < \rho < \rho_2. \quad (12)$$

Above, the dashes denote derivatives with respect to ρ and

$$\begin{aligned} \mathcal{P}(\rho) &:= -\frac{2}{\rho}, & \mathcal{Q}(\rho) &:= \left[\frac{3}{\rho^2} - \mu^2 \left(\omega_0^2 \rho^2 + \frac{1}{\omega_0^2 \rho^2} \right) \right], \\ \mathcal{R}(\rho) &:= -\frac{1}{\rho} \left[\frac{3}{\rho^2} + \mu^2 \left(\omega_0^2 \rho^2 - \frac{3}{\omega_0^2 \rho^2} \right) \right], & \mathcal{S}(\rho) &:= \mu^4. \end{aligned}$$

Note that the inner and, respectively, the outer curved surfaces of the current configuration become

$$\rho_{1,2} = \frac{1}{\omega_0} \left[(1 + 4\eta^2 \omega_0^2)^{1/2} \pm 2\eta \omega_0 \right]^{1/2}, \quad (13)$$

while the principal stretch in the \mathbf{e}_θ -direction assumes the simple form

$$\lambda = \omega_0 \rho. \quad (14)$$

The solution of (12) is found subject to the non-dimensional boundary conditions obtained from (9) via (10),

$$\Phi''' - \mu^2 \left(\omega_0^2 \rho^2 + \frac{2}{\omega_0^2 \rho^2} \right) \Phi' + \frac{\mu^2}{\rho} \left(\omega_0^2 \rho^2 + \frac{2}{\omega_0^2 \rho^2} \right) \Phi = 0, \quad \text{for } \rho = \rho_{1,2}, \quad (15a)$$

$$\Phi'' - \frac{1}{\rho} \Phi' + \frac{\mu^2}{\omega_0^2 \rho^2} \Phi = 0, \quad \text{for } \rho = \rho_{1,2}. \quad (15b)$$

The normal-mode approach has reduced the bifurcation analysis to the study of a standard ordinary eigenproblem for $\Phi(\rho)$ and $\omega_0 \in (0, \pi)$. While for structural mechanics problems (e.g., [23]) this route is free of pitfalls, in finite elasticity it is only deceptively so. The danger is that the bent block might develop shear bands or other material instabilities before the compressed inner surface starts to wrinkle. Such occurrences are heralded by a loss of ellipticity in the partial differential equation (8); unfortunately, they remain undetected by (12). Conveniently, the use of a neo-Hookean constitutive law precludes any form of material instabilities (note that \mathcal{L}_4 is strongly elliptic in this case). Such exotic effects, however, were accounted for in [8], but it was found that the surface instability was always the first to occur.

3 Numerical experiments

The stability of the bent rubber block is now investigated numerically, the starting point being the eigenproblem (12,15) formulate in §2. Our first objective is to find out the dependence of the critical bending angle ω_0 in terms of the aspect ratio $\eta \equiv A/L$. It is expected that an Euler-type buckling instability is experienced for $0 < \eta < \infty$, but in the limit $\eta \rightarrow \infty$ this behaviour degenerates into a surface instability. Such behaviour is consistent with the results of Dryburgh & Ogden [12] and Haughton [11], although it appears that earlier investigators [8] reported only surface instabilities for some other choices of constitutive behaviour. Unfortunately, a direct comparison with those results is not possible here but, intuitively, one would expect that the finite thickness of the block should set a length-scale for the instability pattern.

A rather peculiar feature of our eigenproblem is the dependence of the mode number m in (11) on the bending angle. In order to identify the former quantity for a given η , we shall plot the principal stretch on the curved inner boundary $\rho = \rho_1$ in terms of this number, the critical value of m being that associated with the largest $\lambda_1 \equiv \lambda(\rho_1, \eta; n)$ when $n \in \mathbb{N}$. This

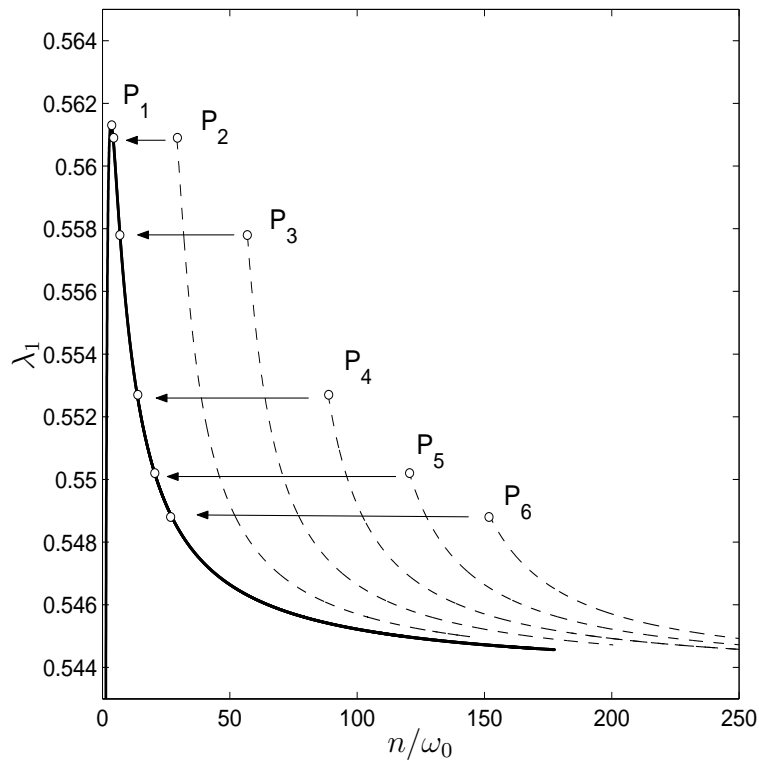


Figure 3: Plot of the critical stretch $\lambda_1 \equiv \lambda(\rho_1)$ against the mode number $m \equiv n/\omega_0$, as obtained by direct numerical integration of the eigenproblem (12,15) for a sample of aspect ratios η . The maximum principal stretch for each individual case considered is marked by a small circle and corresponds to the points: P_1 ($\eta = 1.0$), P_2 ($\eta = 3.0$), P_3 ($\eta = 5.0$), P_4 ($\eta = 10.0$), P_5 ($\eta = 15.0$), and P_6 ($\eta = 20.0$). These maxima are attained for $n = 1$, except for P_1 which corresponds to $n \approx 2.43$.

procedure is carried out in Figure 3 where we consider a sample of values for η (see the caption for details): the horizontal axis records the mode number n/ω_0 while the vertical axis shows λ_1 . Strictly speaking, $n \in \mathbb{N}$ but we shall take this parameter to be a positive real number and notice that the eigenvalue of the problem (12,15), ω_0 , will depend on this quantity as well as on η , i.e., $\omega_0 \equiv \omega_0(\eta, n)$. For $\eta = 1$ we find the curve shown with a continuous line and which consists of two sloping parts separated by a peak, P_1 (corresponding to $n \approx 2.43$). Henceforth, we shall refer to this curve as \mathcal{C}_1 . Note that the right-hand part is monotonic decreasing and unbounded but here only a segment of that curve is shown. The neutral stability curves for the other values of $\eta = \eta_j > 1$ considered are $\mathcal{C}_j := \{(\lambda_1(\rho_1; \eta_j, n), n/\omega_0) \mid n \in \mathbb{R}_+\}$ and they all turn out to be part of \mathcal{C}_1 . The remark made above regarding \mathcal{C}_1 applies for these curves as well. For the sake of clarity in Figure 3 the curves are shifted and shown separately as dashed lines, but their top endpoints ($P_2 \div P_6$) are marked on \mathcal{C}_1 as well. All such points correspond to the choice $n = 1$.

The feature illustrated in Figure 3 is generic and not restricted to the particular values of $\eta > 1$ chosen. A first observation is that the number of ripples on the compressed side of the block increases with the non-dimensional thickness η . When this latter quantity is reasonably large ($\eta > \approx 3$) the critical mode number given by (11) always corresponds to $n = 1$. Thus, the behaviour of a very thick block can be understood in two different ways: (i) assuming that $n = 1$ and $\eta \gg 1$ or, conversely, (ii) fixing $\eta = \mathcal{O}(1)$ and letting $n \gg 1$. In the former case the *critical* mode number will simply be $\pi/\omega(\eta, 1)$, whereas in the latter one the following

observation helps: if $\eta_1, \eta_2 > 0$ are two given, sufficiently large aspect ratios with $\eta_1 < \eta_2$, then

$$\frac{1}{\omega_0(\eta_2, 1)} = \frac{\eta_2}{\omega_0(\eta_1, \eta_2)}.$$

Comparing this with (11) assertion (ii) should now be obvious.

It is instructive to gain some insight into the behaviour of the critical eigenfunctions associated with the P_j 's marked on Figure 3. This information is included in Figure 4 where, for the sake of brevity, we show the radial and azimuthal displacements only for $P_1 \div P_4$, as obtained from the two equations in (7). The localisation of the deformation near the curved

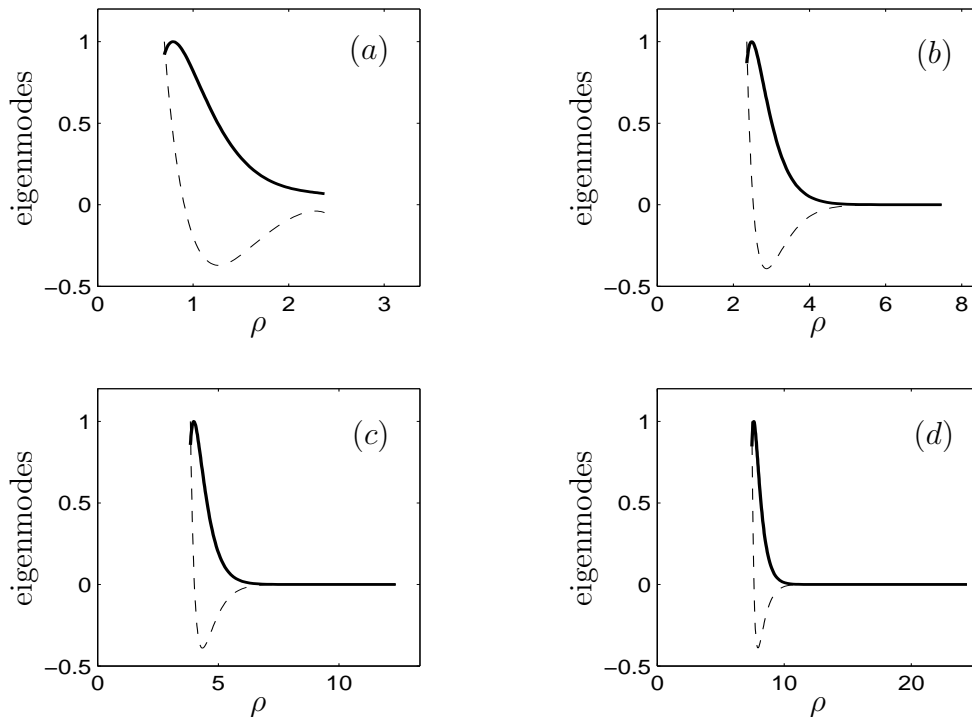


Figure 4: The eigenfunctions associated with the critical points P_j in Figure 3: (a) P_1 ($\eta = 1.0$), (b) P_2 ($\eta = 3.0$), (c) P_3 ($\eta = 5.0$), and (d) P_4 ($\eta = 10.0$). In each plot the continuous line denotes $\rho^{-1}\Phi(\rho)$ (*radial displacement*), while the dashed line is used for $\Phi'(\rho)$ (*azimuthal displacement*). The range for these functions is $\rho_1 \leq \rho \leq \rho_2$ and they are suitably normalised so that their maximum amplitude is unity.

inner surface of the bent block when η increases is clearly obvious. The stress concentration phenomenon revealed by these plots is to be expected because the thicker the rubber block, the more difficult is to bend it, that is, the instability will be likely to occur for small values of the bending angle. Hence, curvature effects will only be “felt” in the immediate proximity of the bent inner surface. In the remaining of the paper we show that this behaviour is ideally suited for a singular perturbation analysis.

At this juncture some remarks on the method used to identify the critical mode number are appropriate. At first sight, our work in Figure 3 might appear a little awkward. The coincidence of the curves \mathcal{C}_j ($j = 2, \dots, 6$) with \mathcal{C}_1 could have been inferred by taking into account that the principal stretch λ_1 is independent of n and depends only on the product $\omega_0\eta$ – see (13) and (14). However, we believe that the longer route taken here has the advantage of

clarifying some of the vague statements made by Haughton in [11]. He misinterpreted the role played by $n \in \mathbb{N}$ in formula (11) and, in Figure 5 of his paper, he varied both n and η ending up with a wrong statement as to the behaviour of the neutral stability curves for the bending problem. For convenience we reproduce that scenario in our Figure 5. Although a different formulation of the eigenproblem was used in [11] (without recourse to any potential function), the results we show are the same (as they should be since the height of the rubber block in [11] was fairly large, $H/A = 10$). The only exception is the unusual feature seen in that paper for $n = 1$, which we did not find with our model. Triantafyllidis [8] seems to have committed a different error by excluding ω_0 from what he refers to as “wave-number” (see Figure 7 in his work). That might explain why he did not find an Euler-type buckling instability.

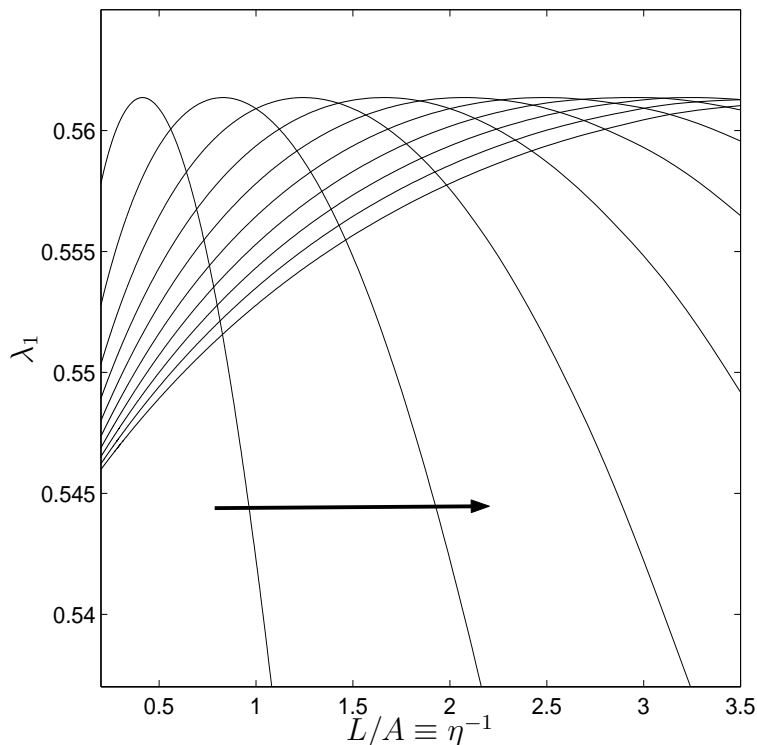


Figure 5: A plot of critical values of $\lambda_1 \equiv \lambda(\rho_1)$ against undeformed length L/A for mode numbers $n = 1 \div 10$ (see also Figure 5 in reference [11]). The arrow indicates the direction of increasing n .

The response curves shown in Figure 5 are reminiscent of similar situations encountered in the buckling of thin-walled structures (e.g., [23]). In that particular type of situation n represents the number of half-waves of the instability pattern and plots like the one shown above can be used to infer the wavelength of the buckling pattern from knowledge of some aspect ratio (related recent work on thin-film instabilities can be found in [1, 2, 3, 4]). However, in the present context such extrapolations appear to provide misleading information for obvious reasons. Also, in the limit $A/L \rightarrow \infty$ the critical principal stretch would have to be equal to the value 0.544 predicted by Biot’s analysis for a neo-Hookean half-plane in compression (cf. [16, 22]). This is clearly not the case in Figure 5 but, on the other hand, the earlier results of Figure 3 do anticipate this expectation.

4 Stress concentration for $n \gg 1$

The mathematical structure of the boundary-value problem derived in §2 is akin to a number of situations investigated recently in the literature by Fu *et al.* [25, 26] and Haughton & Chen [27]. Broadly speaking, these authors encountered a particular occurrence of turning points (e.g., [28]) or repeated roots in the characteristic equations associated with bifurcation analyses for everted cylindrical/spherical shells. It was stated that such special points could aid in detecting sites of high-stress concentration within elastic solids. Furthermore, in light of the recent work on edge-buckling of thin films [1, 2, 3, 4], it would appear very reasonable to conclude that Fu's observation might go a long way towards explaining the localised behaviour seen in Figure 4. That this is not true we are going to see in §4.2, but before we pursue those issues it is important to gain an understanding of the relevance of WKB techniques in the present context. According to the previous interpretation of the parameters $\eta \equiv A/L$ and n (defined in equation (11)), the bifurcation of a thick rubber block ($\eta \gg 1$) can be understood by taking $\eta = 1$ and allowing $n \gg 1$. This is precisely what we do in the remainder of the paper.

4.1 WKB approach

The WKB method is a simple and efficient tool for dealing with variable-coefficient linear differential equations containing certain small or large parameters. We shall exploit the presence of $\mu \equiv \pi n \gg 1$ in our eigenproblem to describe the dependence of ω_0 (or λ_1) on this large parameter.

A WKB solution of (12) is sought in the form

$$\Phi(\rho) = Y(\rho) \exp\left(\mu \int_{\rho_1}^{\rho} S(\xi) d\xi\right), \quad (16a)$$

$$Y(\rho) = Y_0(\rho) + \frac{1}{\mu} Y_1(\rho) + \frac{1}{\mu^2} Y_2(\rho) + \dots \quad (16b)$$

where $S \equiv S(\rho)$ is one of the roots of the characteristic equation

$$S^4 - \left(\omega_0^2 \rho^2 + \frac{1}{\omega_0^2 \rho^2}\right) S^2 + 1 = 0,$$

and $Y_j(\rho)$ ($j = 0, 1, \dots$) are functions that are to be determined sequentially by substituting the ansatz (16) into (12), and then solving the differential equations obtained by setting to zero the coefficients of like powers of μ . The above bi-quadratic has four real roots that will be labelled

$$S_1^{(\pm)}(\rho) := \pm \omega_0 \rho, \quad S_2^{(\pm)}(\rho) := \pm \frac{1}{\omega_0 \rho},$$

and they lead to a set of linearly independent (approximate) solutions for (12). Given our experience with the direct numerical simulations of §3, it is expected that only the exponentials corresponding to $S_{1,2}^{(-)}(\rho)$ need to be used in order to capture the through-thickness localised behaviour. The superscripts “1” and “2” will be used to identify quantities associated with these characteristic exponents in (16).

The determinantal equation that follows by imposing the boundary conditions (15) at $\rho = \rho_1$ on the WKB solutions $\Phi^{(1)}$ and $\Phi^{(2)}$ has the form

$$U_1(\rho_1)V_2(\rho_1) - U_2(\rho_1)V_1(\rho_1) = 0, \quad (17)$$

where

$$U_j(\rho) := \Phi^{(j)'''} - \mu^2 \left(\omega_0^2 \rho^2 + \frac{2}{\omega_0^2 \rho^2} \right) \Phi^{(j)'} + \frac{\mu^2}{\rho} \left(\omega_0^2 \rho^2 + \frac{2}{\omega_0^2 \rho^2} \right) \Phi^{(j)}, \quad (18)$$

$$V_j(\rho) := \Phi^{(j)''} - \frac{1}{\rho} \Phi^{(j)'} + \frac{\mu^2}{\omega_0^2 \rho^2} \Phi^{(j)}, \quad j = 1, 2. \quad (19)$$

When calculating $\Phi^{(j)}$ ($j = 1, 2$) we shall ignore terms of order $\mathcal{O}(\mu^{-2})$ and higher in the ansatz (16b). These solutions are fixed by routinely solving a series of non-homogeneous linear differential equations. The various multiplicative and additive constants in the expressions of those functions can be chosen (without loss of generality) to be unity or, respectively, equal to zero. The final results are

$$\begin{aligned} Y_0^{(1)}(\rho) &= \frac{\rho}{(1 - \omega_0^4 \rho^4)^{1/2}}, & Y_1^{(1)}(\rho) &= -\frac{5\omega_0^8 \rho^8 + 10\omega_0^4 \rho^4 - 3}{4\omega_0 \rho^2 (1 - \omega_0^4 \rho^4)^2} Y_0^{(1)}(\rho), \\ Y_0^{(2)}(\rho) &= \frac{\rho^2}{(1 - \omega_0^4 \rho^4)^{1/2}}, & Y_1^{(2)}(\rho) &= \frac{3\omega_0(\omega_0^4 \rho^4 + 1)}{2(1 - \omega_0^4 \rho^4)^2} Y_0^{(2)}(\rho), \end{aligned}$$

and thus,

$$\Phi^{(j)}(\rho) \approx \left\{ Y_0^{(j)}(\rho) + \frac{1}{\mu} Y_1^{(j)}(\rho) \right\} \exp \left(\mu \int_{\rho_1}^{\rho} S_j^{(-)}(\xi) d\xi \right), \quad j = 1, 2. \quad (20)$$

It must be noted that $Y_j^{(j+1)}$ ($j = 0, 1$) blow up when $\rho = \bar{\rho} \equiv \omega_0^{-1} \in (\rho_1, \rho_2)$; in the language of differential equations this represents a (multiple) turning point of the differential equation (12). Such points are usually defined as those values of the independent variable for which some of the roots of the characteristic equation coalesce. For this particular example, both $S_1^{(+)}(\bar{\rho}) = S_2^{(+)}(\bar{\rho})$ and $S_1^{(-)}(\bar{\rho}) = S_2^{(-)}(\bar{\rho})$, *i.e.*, two pairs of roots merge. Strictly speaking, the validity of the above formulae for $Y_j^{(j+1)}$ ($j = 0, 1$) requires $|\rho - \bar{\rho}| \gg \mu^{-1/2}$. Although $\bar{\rho}$ depends on the unknown eigenvalue, our numerical experiments suggest that the turning point remains confined to the central part of the interval (ρ_1, ρ_2) .

On making use of (20) into (17), we are able to expand the determinantal equation in decreasing integral powers of $\mu \gg 1$,

$$\Gamma_0(\lambda_1) + \Gamma_1(\omega_0, \lambda_1) \frac{1}{\mu} + \Gamma_2(\omega_0, \lambda_1) \frac{1}{\mu^2} + \dots = 0, \quad (21)$$

with

$$\begin{aligned} \Gamma_0(z) &:= 8z^2(z^2 + 1)^2(z^3 - z^2 + z + 1)(z^3 + z^2 + z - 1), \\ \Gamma_1(\omega_0, z) &:= 2\omega_0(4z^{12} + 15z^{10} + 23z^8 + 8z^4 - 7z^2 - 3), \end{aligned}$$

and

$$\begin{aligned} \Gamma_2(\omega_0, z) &:= \frac{\omega_0^2}{(z^2 + 1)^2(z^2 - 1)^4} (6z^{22} - 18z^{20} - 37z^{18} - 477z^{16} - 1118z^{14} \\ &\quad - 930z^{12} - 600z^{10} - 4z^8 - 120z^6 - 140z^4 + 45z^2 + 33). \end{aligned}$$

The solution of (21) yields approximations for both the critical bending angle ω_0 , and the principal stretch λ_1 . For the sake of brevity we record only the final results here

$$\omega_0 = \bar{\Omega}_0 + \frac{\bar{\Omega}_1}{\mu} + \frac{\bar{\Omega}_2}{\mu^2} + \dots, \quad (22)$$

$$\bar{\Omega}_0 = 0.771844, \quad \bar{\Omega}_1 = -1.305565, \quad \bar{\Omega}_2 = 15.39664,$$

and

$$\lambda_1 = \Lambda_0 + \frac{\Lambda_1}{\mu} + \frac{\Lambda_2}{\mu^2} + \dots, \quad (23)$$

$$\Lambda_0 = 0.543689, \quad \Lambda_1 = 0.385922, \quad \Lambda_2 = -4.184333.$$

As one would expect, $\Lambda_0 \approx 0.544$ represents the critical value of the principal stretch for the surface instability of a compressed neo-Hookean half-space (cf. [22]); the next-order corrections in formula (23) account for the finite size of the rubber block. To assess the usefulness of the two asymptotic results (22) and (23), a set of comparisons with direct numerical simulations is recorded in Table 1. The agreement is excellent for both ω_0 and λ_1 ; in particular, we find that the relative accuracy (RA) associated with ω_0 ranges between 1.4% ($n = 7$) and 0.8% ($n = 20$). The approximation of λ_1 is even better, for RA is at most 0.4% ($n = 7$) in all cases considered.

The WKB analysis laid out above has the advantage of producing a robust approximation for ω_0 (or λ_1) with minimum effort. The presence of the turning point, however, is worrying because it tends to obscure the true nature of the localised behaviour exhibited by (12). It is not immediately clear whether such behaviour has anything to do with the turning point and, thus, a change of tack is imperative. It will shortly become obvious that conventional boundary-layer techniques are better suited for understanding the underlying mathematical structure responsible for the scenario depicted in Figure 4. The details of that particular approach are highlighted next.

4.2 Boundary-layer analysis

To begin, we introduce the stretched variable $X = \mathcal{O}(1)$ such that $\rho = \rho_1 + X\mu^{-1}$, and look for solutions of (12) with

$$W(X) = W_0(X) + W_1(X)\frac{1}{\mu} + W_2(X)\frac{1}{\mu^2} + \dots, \quad (24a)$$

$$\omega_0 = \Omega_0 + \frac{\Omega_1}{\mu} + \frac{\Omega_2}{\mu^2} + \dots, \quad (24b)$$

$$\rho_1 = \Delta_0 + \frac{\Delta_1}{\mu} + \frac{\Delta_2}{\mu^2} + \dots. \quad (24c)$$

Although (24b) and (24c) are not independent, it helps to expand ρ_1 in the form suggested here. Of course, when solving the governing equations for the coefficients $W_j(X)$ ($j = 0, 1, \dots$) one has to remember formula (13) and replace the Δ_k 's with their expressions in terms of the Ω_j ($j = 0, 1, \dots, k$).

On substituting (24) into (12) we find a hierarchy of differential equations

$$\mathcal{L}_{\text{BL}}[W_k] = \sum_{i=0}^{k-1} \sum_{j=1}^3 A_{ij}^{(k)} \frac{d^j W_i}{dX^j} \quad (k \geq 0), \quad (25)$$

Table 1: Comparisons between direct numerical simulations of the eigenproblem (12,15) and the asymptotic results recorded in the formulae (22) and (23). The bending angle and the azimuthal stretch on the inner boundary associated with the former set of values are denoted by ω_0^{num} and, respectively, λ_1^{num} . The corresponding asymptotic quantities are identified as ω_0^{asy} and λ_1^{asy} .

n	ω_0^{num}	ω_0^{asy}	λ_1^{num}	λ_1^{asy}
7	0.744313103	0.733652725	0.555301084	0.552585690
8	0.744272309	0.736778030	0.554325988	0.552419947
9	0.744928414	0.739453964	0.553494699	0.552104104
10	0.745886633	0.741762924	0.552780068	0.551733662
11	0.746957132	0.743772049	0.552160229	0.551352710
12	0.748046186	0.745533828	0.551618218	0.550981720
13	0.749107553	0.747090530	0.551140475	0.550629796
14	0.750119338	0.748474960	0.550716526	0.550300415
15	0.751072409	0.749714126	0.550337796	0.549994245
16	0.751964384	0.750829238	0.549997574	0.549710574
17	0.752796399	0.751837997	0.549690282	0.549448050
18	0.753571374	0.752754745	0.549411416	0.549205075
19	0.754293016	0.753591537	0.549157200	0.548980000
20	0.754965302	0.754358139	0.548924583	0.548771235

in which

$$\mathcal{L}_{\text{BL}} := \frac{d^4}{dX^4} - \left(\zeta_0^2 + \frac{1}{\zeta_0^2} \right) \frac{d^2}{dX^2} + 1$$

is the boundary-layer (BL) differential operator and $\zeta_0 := \Delta_0 \Omega_0$. The quantities $A_{ij}^{(k)} \equiv A_{ij}^{(k)}(X)$ will be defined as we go along, with the convention that $A_{ij}^{(0)} \equiv 0$. In contrast to the WKB analysis the general solution of each one of the equations in (25) is trivially found, for \mathcal{L}_{BL} has constant coefficients.

The equations (25) are solved subject to two types of boundary conditions. The first set is obtained from (15) with the help of the ansatz (24), and can be cast in the general form

$$\mathcal{H}_1[W_k] = \sum_{i=0}^{k-1} \sum_{j=0,1} B_{ij}^{(k)} \frac{d^j W_i}{dX^j}, \quad \text{for } X = 0, \quad (26a)$$

$$\mathcal{H}_2[W_k] = \sum_{i=0}^{k-1} \sum_{j=0,1} C_{ij}^{(k)} \frac{d^j W_i}{dX^j}, \quad \text{for } X = 0, \quad (26b)$$

where

$$\mathcal{H}_1 := \frac{d^3}{dX^3} - \left(\zeta_0^2 + \frac{2}{\zeta_0^2} \right) \frac{d}{dX} \quad \text{and} \quad \mathcal{H}_2 := \frac{d^2}{dX^2} + \frac{1}{\zeta_0^2};$$

the remark made for the $A_{ij}^{(k)}$'s applies to the boundary coefficients $B_{ij}^{(k)}$ and $C_{ij}^{(k)}$ as well.

The second set of boundary conditions is motivated by the numerical experiments illustrated in Figure 4 and involves the requirement that

$$\frac{d^j W_i}{dX^j} \rightarrow 0 \quad \text{as } X \rightarrow \infty, \quad (27)$$

for $i \geq 0$, $j = 0, \dots, 3$. It might also be worth pointing out that the solution of equation (12) is exponentially small in the outer layer (cf. §4.1), so that (27) can be viewed as matching conditions between the inner and the outer solutions.

The leading order problem for $W_0(X)$ is homogeneous and consists of the differential equation (25) for $k = 0$, together with the boundary conditions (26). Rejecting the exponentially growing contributions and imposing the normalisation condition $W_0(X = 0) = 1$, it follows that

$$W_0(X) = \frac{2}{1 - \zeta_0^4} \exp(-\zeta_0 X) - \frac{1 + \zeta_0^4}{1 - \zeta_0^4} \exp(-X/\zeta_0). \quad (28)$$

When this function is substituted into the boundary conditions, one obtains an algebraic equation,

$$\zeta_0^8 + 2\zeta_0^4 - 4\zeta_0^2 + 1 = 0,$$

whose only acceptable solution is $\zeta_0 \approx 0.543689$. The result is identical to Λ_0 found in §4.1 and the same turns out to be true for Ω_0 in (24b).

The next order problem corresponds to taking $k = 1$ in (25) and (26). The coefficients that appear in these equations are

$$\begin{aligned} A_{01}^{(1)} &:= \frac{1}{\Delta_0 \zeta_0^2} (\zeta_0^4 - 3), & A_{02}^{(1)} &:= \frac{2}{\zeta_0^3} (\Delta_0 \Omega_1 + \Delta_1 \Omega_0 + \Omega_0 X) (\zeta_0^4 - 1), & A_{03}^{(1)} &:= \frac{2}{\Delta_0}, \\ B_{00}^{(1)} &:= -\frac{\Omega_0}{\zeta_0^3} (\zeta_0^4 + 2), & B_{01}^{(1)} &:= \frac{2}{\zeta_0^3} (\Delta_0 \Omega_1 + \Delta_1 \Omega_0) (\zeta_0^4 - 2), \\ C_{00}^{(1)} &:= \frac{2}{\zeta_0^3} (\Delta_0 \Omega_1 + \Delta_1 \Omega_0), & C_{01}^{(1)} &:= \frac{1}{2} A_{03}^{(1)}. \end{aligned}$$

The first-order correction Ω_1 in the expansion (24b) of the eigenvalue ω_0 is recovered by enforcing the Fredholm solvability condition on the *non-homogeneous* problem satisfied by $W_1(X)$. The task is simplified by the observation that the homogeneous problem for $W_0(X)$ is self-adjoint. Standard calculations show that this constraint amounts to

$$\begin{aligned} C_{01}^{(1)} \left\{ \frac{dW_0}{dX}(0) \right\}^2 + \left\{ C_{00}^{(1)} - B_{01}^{(1)} \right\} W_0(0) \frac{dW_0}{dX}(0) - B_{00}^{(1)} \{W_0(0)\}^2 \\ = \int_0^\infty W_0(\xi) \sum_{j=1}^3 A_{0j}^{(1)} \frac{d^j W_0}{dX^j}(\xi) d\xi. \quad (29) \end{aligned}$$

Notice that the integral on the right-hand side in (29) is evaluated analytically, and thus the solvability condition will reduce to a linear equation in Ω_1 . The solution $\Omega_1 \approx -1.305562$ is, for all practical purposes, identical to the WKB result obtained earlier.

The pattern of the boundary-layer approach for the pure bending problem is now clear: at each step one will have to impose a solvability condition for finding Ω_j that features in (24b), and then solve (exactly) a non-homogeneous fourth-order boundary-value problem in order to get $W_j(X)$. The algebraic manipulations become increasingly unwieldy as we move to further orders, but symbolic algebra packages help considerably. We have imposed the solvability condition for the W_2 -problem and found that the value of Ω_2 predicted agrees with $\bar{\Omega}_2$ to within five significant digits; for completeness, the coefficients needed to set up that problem are recorded below

$$A_{11}^{(2)} := \frac{\Omega_0}{\zeta_0^3} (\zeta_0^4 - 3), \quad A_{12}^{(2)} := A_{02}^{(1)}, \quad A_{13}^{(2)} := A_{03}^{(1)},$$

$$\begin{aligned}
 A_{01}^{(2)} &:= \frac{1}{\zeta_0^4} \left[\Omega_0^2 (X + \Delta_1) (\zeta_0^4 + 9) + 2\Omega_1 (\zeta_0^4 + 3) \zeta_0 \right], & A_{03}^{(2)} &:= -\frac{2}{\Delta_0^2} (X + \Delta_1), \\
 A_{02}^{(2)} &:= \frac{1}{\zeta_0^4} \left\{ 2\Delta_0 \Omega_0 (\Delta_0 \Omega_2 + \Delta_2 \Omega_0) (\zeta_0^4 - 1) + [\Omega_0^2 (X + \Delta_1)^2 + \Delta_0^2 \Omega_1^2] (\zeta_0^4 + 3) \right. \\
 &\quad \left. + 4\Omega_1 (X + \Delta_1) (\zeta_0^4 + 1) \zeta_0 - 3\Omega_0^2 \zeta_0^2 \right\}, \\
 B_{00}^{(2)} &:= -\frac{1}{\zeta_0^4} \left[\Delta_1 \Omega_0^2 (\zeta_0^4 - 6) + 2\Omega_1 (\zeta_0^4 - 2) \zeta_0 \right], \\
 B_{01}^{(2)} &:= \frac{1}{\zeta_0^4} \left[2(\Delta_0 \Omega_2 + \Delta_2 \Omega_0) (\zeta_0^4 - 2) \zeta_0 + (\Delta_0^2 \Omega_1^2 + \Delta_1^2 \Omega_0^2) (\zeta_0^4 + 6) + 4\Delta_1 \Omega_1 (\zeta_0^4 + 2) \zeta_0 \right], \\
 B_{10}^{(2)} &:= B_{00}^{(1)}, & B_{11}^{(2)} &:= B_{01}^{(1)}. \\
 C_{00}^{(2)} &= -\frac{1}{\zeta_0^4} \left[3(\Delta_0^2 \Omega_1^2 + \Delta_1^2 \Omega_0^2) - 2\zeta_0 (\Omega_0 \Delta_2 + \Delta_0 \Omega_2 - 2\Delta_1 \Omega_1) \right], \\
 C_{01}^{(2)} &= -\frac{\Delta_1}{\Delta_0^2}, & C_{10}^{(2)} &= C_{00}^{(1)}, & C_{11}^{(2)} &= C_{01}^{(1)}.
 \end{aligned}$$

It might appear that our boundary-layer approach is a by-product of adopting the simple neo-Hookean form for the constitutive response of the bulk material. However, this impression is only apparent, for choosing a strain-energy function of the form

$$W(\lambda_r, \lambda_\theta) \propto (\lambda_r^q + \lambda_\theta^q), \quad (q > 0),$$

the same mathematical structure persists. In this case the interpretation of the asymptotic results become more difficult because now loss of ellipticity will be encountered for some value of $q \neq 2$; such issues will be discussed elsewhere [29].

5 Concluding remarks

We have re-examined the bifurcations in cylindrical bending of a thick rubber block under the assumption of plane-strain deformation. The constitutive behaviour was taken to be that of a neo-Hookean incompressible solid, the reason for this being twofold: we wanted to (a) exclude any material instabilities and, (b) simplify our equations as much as possible. The outcome turned out to be a simple fourth-order eigenproblem with variable coefficients. Direct numerical simulations and singular perturbation methods were employed to unravel the origins of the rippling pattern triggered on the compressed face of the block, when the bending angle is sufficiently large. It has been shown in §3 that previous investigators [8, 11] misinterpreted the definition of the so-called “mode number” and thus made several erroneous statements. In particular, we want to re-iterate here that blocks of a large but *finite thickness* will always experience an Euler-type buckling instability with a well defined number of ripples. It is only in the limit of an *infinitely thick* block that one finds the degenerate surface instability. Our results deal with the neo-Hookean material, but it is believed that the above statement remains valid in other cases as well. It seems quite unlikely that the choice of constitutive law would have any dramatic repercussions on the overall arguments presented in this paper.

Our work has also demonstrated that the use of WKB methods in the context of incremental elasticity is unnecessary and even misleading. The multiple turning point featuring

in the present eigenproblem has little to do with the tendency of the rippling deformation to confine itself near the inner curved surface of the block. Interestingly though, a simple-minded boundary-layer analysis was able to expose the nature of the localisation in a trivial way. In spite of the original complexity of the problem, we found that if the rubber block is sufficiently thick, its possible bifurcations from the cylindrical configuration are governed by constant-coefficient differential equations – easily solvable in closed form. This has important overall implications since preliminary calculations indicate that the problems taken up in [25, 26, 27]) are amenable to a similar boundary-layer analysis. We shall report the corresponding details in a forthcoming study [29].

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